



PLANCK Low Frequency Instrument: towards a final imaging of the CMB anisotropies

M. Sandri¹, M. Bersanelli², C. Burigana¹, R. C. Butler¹, F. Cuttaia¹, F. Finelli¹,
E. Franceschi¹, A. Gruppuso¹, M. Malaspina¹, N. Mandolesi¹, A. Mennella³,
G. Morgante¹, G. Morigi¹, L. Popa¹, L. Terenzi¹, L. Valenziano¹ and F. Villa¹

¹ Istituto IASF CNR/INAF, Sezione di Bologna, Via Gobetti, 101, I-40129 Bologna, Italy
e-mail: sandri@bo.iasf.cnr.it

² Dipartimento di Fisica, Università degli Studi di Milano, Via Celoria, 16, I-20133 Milano, Italy
e-mail: marco.bersanelli@fisica.unimi.it

³ Istituto IASF CNR/INAF, Sezione di Milano, Via Bassini, 15, I-20133 Milano, Italy
e-mail: daniele@mi.iasf.cnr.it

Abstract. PLANCK is the third generation of mm-wave instruments designed for space observations of the cosmic microwave background (CMB) anisotropies within the new Cosmic Vision 2020 ESA Science Program. PLANCK will map the whole sky with unprecedented sensitivity, angular resolution, and frequency coverage, and it likely leads us to the final comprehension of the CMB anisotropies. The Low Frequency Instrument (LFI), operating in the 30 ÷ 70 GHz range, is one of the two instruments onboard PLANCK satellite, sharing the focal region of a 1.5 meter off-axis dual reflector telescope together with the High Frequency Instrument (HFI) operating at 100 ÷ 857 GHz. We present LFI and discuss the major instrumental systematic effects that could degrade the measurements and the solutions adopted in the design phase in order to adequately reduce and control them. (*On behalf of LFI Consortium*)

Key words. Space mission, experimental cosmology, cosmic microwave background

1. Introduction

PLANCK represents the third generation of mm-wave instruments designed for space observations of the cosmic microwave background (CMB) anisotropies within the new Cosmic Vision 2020 ESA Science Program. Following the present NASA's mission Wilkinson Microwave Anisotropy Probe (Bennet et al. 2003), PLANCK will map the whole sky with unprecedented sensitivity

(the average sensitivity on a pixel of FWHM side in the measurement of the temperature anisotropy is about two parts per million), angular resolution (FWHM from about 30 down to 5 arcmin), and frequency coverage, and it likely leads us to the final comprehension of the CMB anisotropies (Burigana et al. 2003, *this issue*). The Low Frequency Instrument (LFI, Mandolesi et al. 1998), operating in the 30 ÷ 70 GHz range, is one of the two instruments onboard PLANCK satellite, sharing

the focal region of a 1.5 meter off-axis dual reflector telescope together with the High Frequency Instrument (HFI, Puget et al. 1998) operating at $100 \div 857$ GHz. LFI consists of four main units: the front end unit (FEU, Sect. 2), the back end unit (BEU, Sect. 3), the radiometer electronics box assembly (REBA, Sect. 4), and the sorption cooler system (SCS, Sect. 5). We describe how these units have been conceived, as well as the major instrumental systematic effects that could degrade the measurements and the solutions adopted in the design phase in order to adequately reduce and control them. In addition, also the LFI data processing center (DPC) has been mentioned and described in Sect. 6.

2. The Front End Unit

2.1. Feed horns and OMTs

LFI is coupled to the PLANCK telescope by an array of conical dual profiled corrugated feed horns (Villa et al. 2002). Dual profiled corrugated horns have been selected as the best design in terms of shape of the main lobe, very low level of cross polarization and side-lobe, control of the phase centre location, low weight and compactness. In addition, the electromagnetic field inside the horn propagates with low attenuation and low return loss. The ortho mode transducers (OMTs) separate the orthogonal polarizations with minimal losses and cross-talk. The straylight, or the unwanted off-axis radiation contributing to the observed signal, represents one of the major systematic effects in PLANCK/LFI and can be controlled optimizing the feed horn design, since from the latter depends the illumination of the mirrors. A trade-off between angular resolution (the more the primary mirror is illuminated, the best is the angular resolution) and straylight rejection (the more the mirrors are illuminated, the worst is the straylight rejection) has been obtained for each feed horn coupled with the PLANCK telescope. This optimization has been performed computing the full optical response of several realistic feed horn patterns (Sandri et al. 2003) and convolving the full pattern with the sky signal by considering the ob-

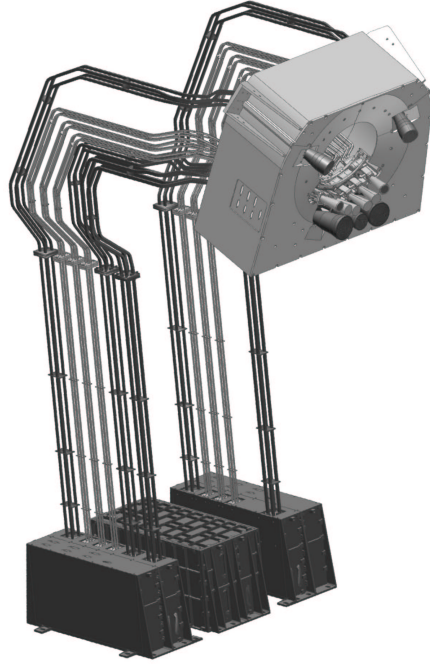


Fig. 1. The Low Frequency Instrument: the front end unit is at the top, whereas the back end unit, the radiometer electronics box assembly, and sorption cooler system are in the lower part.

servational strategy (Burigana et al. 2003), in order to calculate the straylight contamination for each feed model analyzed. As a result, the LFI feed horns have different designs (i.e. inner corrugation profile), depending from their location on the focal surface, and the corresponding angular resolution achieved is the best one, satisfying the straylight rejection requirement.

2.2. Hybrids, phase switches and amplifiers

The OMTs are followed by blocks containing hybrid couplers and amplifiers (including phase switches and output hybrids), all cooled to 20K by the H₂ sorption cooler system. This front-end is designed to minimize the $1/f$ noise in the radiometer (one of the most important potential source of systematic effects) while maintaining low thermal noise

(Seiffert et al. 2002; Mennella et al. 2003). Each block contains two hybrid couplers: each hybrid has two inputs, one of which sees the sky, the other one looks at the 4K reference load through a small rectangular horn. The hybrid coupler combines the signals from the sky and cold load with a fixed phase offset of either 90° or 180° between them. It has the necessary bandwidth, low loss, and amplitude balance needed at the output to ensure adequate signal isolation. The low-noise amplifiers use InP HEMTs in cascaded gain stages. Of all transistors, InP HEMTs have the highest frequency response, lowest noise, and lowest power dissipation. The amplifiers at 30 and 44 GHz use discrete InP HEMTs incorporated into a microwave integrated circuit (MIC). At these frequencies, cryogenic MIC amplifiers have demonstrated noise figures of about 10K, with 20% bandwidth. At 70 GHz, MMICs (Monolithic Microwave Integrated Circuits) architectures, which incorporate all circuit elements and the HEMT transistors on a single InP chip, are used. The LFI will fully exploit both MIC and MMIC technologies at their best. For all frequencies, 30–40 dB of gain are sufficient to guarantee that the overall noise is dominated by the front end amplifiers. If additional gain were located in the front end, the power dissipated would grow significantly, putting too much load on the cooler. Following amplification the signals are passed through a phase switch. The switch consumes microwatts of power, it is broad band, and it works at cryogenic temperatures with switch rates in excess of 1 kHz. The phase switch adds 90° or 180° of phase lag to the signals, thus selecting the input source as either the sky or the reference load at the radiometer output. The phase lagged pair of signals is then passed into a second hybrid coupler, separating the signals. The signals are then transitioned to high performance two meter long bent twisted composite (copper – stainless steel – gold-plated stainless steel) rectangular waveguides that carry the double chain signal to the BEU. The waveguides are thermally connected to the three V-grooved shields of the payload at 50, 90, and 140K from the FEU to the BEU, respectively.

3. The Back End Unit

3.1. Back end modules

Each back end module (BEM) comprises two parallel chains of amplification, filtering, detection, and integration. The detected signals are amplified and a low-pass filter reduces the variance of the random signal, providing in each channel a DC output voltage related to the average value. Post-detection amplifiers are integrated into the BEMs to avoid data transmission problems between the radiometer and the electronics box. The sky and reference signals are at different levels, which are equalized after detection and integration by modulating the gain synchronously with the phase switch. Because of the phase switching in the front end modules, a given detector alternately sees the sky and the reference signals. Differences between the detectors are therefore common-mode in the output signals and have no effect on the final difference, which is calculated in the signal processing unit. Each back end module will be packaged, including analog-to-digital converters, into a box of a few centimeters on a side, including the biasing circuitry and the input and output connectors.

3.2. Data acquisition electronics

The data acquisition electronics (DAE) comprises the analog conditioning electronics, the multiplexers, the analog-to-digital converters, the parallel-to-serial converters, the control electronics, the communication interface, and the power conditioning and distribution electronics. It performs the following functions: communication with the data processing unit (DPU), including command reception and status transmission; acquisition, conditioning, and multiplexing of the signals; control of the data acquisition chain; transmission of raw data to the signal processing unit (SPU); power supply conditioning and distribution; DC biasing of the FEU and BEU amplifiers; synchronous control of the FEU phase switches; and ON/OFF control of FEU and BEU amplifiers.

4. The REBA

The radiometer electronics box assembly (REBA) comprises three sub-units: the SPU, the DPU, and the power supply unit (PSU). The REBA supplies all the telemetry and telecommand communication interfaces with the spacecraft, controls the radiometer array assembly through its interface to the BEU, and processes all the radiometer outputs which have been analogue to digital converted in the BEU into science telemetry.

5. The H₂ Sorption Cooler

The LFI FEU is cooled to 20K by the hydrogen sorption cooler developed at the JPL (NASA). The operating cooler also provides 18K pre-cooling to the HFI 4K cooler. Each cooler is a Joule-Thomson cooler in which 0.0065 g/s of hydrogen expands from 5 MPa to 0.03 MPa through a Joule-Thomson (J-T) expander. The high and low gas pressures are maintained by the fact that the equilibrium pressure of gas above the sorbent bed is a strong function of temperature.

6. The LFI DPC

From the mission operation center (MOC, that will control the PLANCK spacecraft), the scientific data produced by PLANCK will be piped daily to two data processing centres (DPCs). These DPCs will be responsible for all levels of processing of the PLANCK data, from raw telemetry to deliverable scientific products. Although contributions to the LFI DPC,

in terms of information on instrument characteristics and prototype software, come from a variety of geographically distributed sites, the operations are mainly centralized in Trieste, where they are run jointly by the OAT and SISSA. The structure of the LFI (and HFI) DPC has been divided into five levels: LS (during the pre-launch phase, simulation of data acquired from the PLANCK mission on the basis of a software system agreed upon across Consortia), L1 (telemetry processing and instrument control), L2 (data reduction and calibration), L3 (component separation and optimization), and L4 (generation of final products).

Acknowledgements. LFI is funded by the national space agencies of the Institutes of the PLANCK Consortium. In particular the Italian participation is funded by ASI.

References

- Bennett, C.L. et al., 2003, ApJS, 148, 1
- Burigana, C. et al., 2003, submitted to A&A, astro-ph/0303645
- Mandolesi, N. et al., 1998, PLANCK/LFI, A Proposal Submitted to the ESA
- Mennella, A. et al., 2003, ESA Proceedings, WPP-212, 69
- Puget J.L. et al., 1998, HFI for the PLANCK Mission, A Proposal Submitted to the ESA
- Sandri, M. et al., 2003, submitted to A&A, astro-ph/0305152
- Seiffert, M. et al., 2002, A&A, 391, 1185
- Villa, F. et al., 2002, Experimental Astronomy, 14, 1